

transition from the stowed state to the extruded state; and the extrusion plate comprising opposing threads to the threaded rod.

4. The antenna of claim 1, wherein the one or more communication patches of the nadir panel are C-band patches.
5. The antenna of claim 4, wherein the nadir panel comprises two of the C-band patches, a first one configured to receive communication signals and a second one to send the communication signals.
6. The antenna of claim 1 wherein the one or more additional communication patches of the one or more petals are S-band patches.
7. The antenna of claim 6, wherein each of the one or more petals comprises four of the one or more additional communication patches.
8. The antenna of claim 1, wherein the antenna comprises eight petals, the eight petals grouped into four pairs of petals, each pair having a receive petal and a send petal.
9. The antenna of claim 1, wherein the one or more additional communication patches are combined into a corporate feed.
10. The antenna of claim 1, further comprising a latch, wherein the extension mechanism comprises a spring coiled around the extrusion support structure, the spring contacting the nadir panel and the extension plate and the latch maintaining the spring in coiled in the stowed state and the extruded state and operable to release the spring in the extended state.
11. The antenna of claim 10, wherein the extrusion support structure comprises the latch, the latch extending from the outer diameter of the extrusion support structure to contact the extension plate.
12. The antenna of claim 10, wherein the extension plate comprises the latch, the latch extending into a notch in the extrusion support structure.
13. The antenna of claim 1, wherein an angle of the one or more petals is between 5 and 45 degrees.
14. The antenna of claim 1, wherein the one or more petals comprise a dielectric material with a dielectric constant between 1 and 3.
15. The antenna of claim 1, wherein the antenna is configured to receive communication signals from a ground terminal with a horizon angle ranging from 0 to 10 degrees.

Description

BACKGROUND

Communications satellites are artificial structures that relay and in some cases amplify telecommunications signals. They create a communication channel between a source transmitter and a receiver at different ground-based location, such as locations on Earth. Communications satellites may be utilized for television, telephone, radio, internet, and military applications. Wireless communication utilizes electromagnetic waves to carry signals. These waves require line-of-sight, and are thus obstructed by the curvature of the Earth, or other objects. The purpose of communications satellites is to relay the signal around the curve of the Earth, or other objects, allowing communication between widely separated points. Communications satellites may be carried into orbit by a carrier craft. Storage space aboard these craft may be expensive. Thus, a design that minimizes the volume

The satellite 104 may have a control system to determine which petals to utilize to communicate with the ground terminal 102 and other ground terminals.

Referring to FIG. 2-FIG. 6, a satellite 200 is depicted in multiple states and comprises a body 202, an antenna 302, and an antenna extension assembly 502. The satellite 200 may have three states: a stowed state, an extruded state, and an extended state, and transitions between those states. In some embodiments, the satellite 200 may transition between the states multiple time as determined by a control system aboard the satellite 200. In other embodiments, the satellite 200 transitions from the stowed state, through the extruded state, to the extended state once per the control system.

The satellite 200 may be stored in the stowed state (FIG. 2) to be launched into orbit by a carrier craft. While in the stowed state, the antenna 302 and the antenna extension assembly 502 are stored within the body 202 of the satellite 200. The body 202 of the satellite 200 may be hollow to store the antenna 302 and the antenna extension assembly 502. The antenna 302 does not extend past the body 202 of the satellite 200. This enables the satellite 200 to minimize the volume when stored on a carrier craft. The satellite 200 may further have panels on the body 202 that may be deployed and folded to minimize the volume on a carrier craft. The antenna 302 may be deployed before, along with, or after the panels. For example, a control system may detect that the panels have been deployed and, in response, initiate the deployment of the antenna 302.

After the deployment of the antenna 302 is initiated, the antenna 302 transitions (FIG. 3) to the extruded state (FIG. 4). The antenna 302 is moved relative to the body 202 of the satellite 200. The transition to the extruded state may be performed through the operation of a motor-driven device or a stored-energy device, such as a spring. The motor-driven method may enable the speed of deployment to be controllable. For example, a motor turning a threaded rod or screw to drive the antenna 302 out of the body 202 of the satellite 200 may control the rate of the rotation of the threaded rod or screw. Further, a control system may determine when the extruded state has been reached by the number of turns of the screw or rod mechanism. The system may then perform the transition to the extended state. For a stored potential method, a latch, or other mechanism, may hold the antenna 302 in place, and release the antenna 302 by the action of the control system.

Once the satellite 200 is in the extruded state (FIG. 4), the satellite 200 may transition (FIG. 5) to the extended state (FIG. 6). A further motor or potential energy system may be utilized. The extension system moves the antenna extension assembly 502 back toward the body 202 of the satellite 200. The antenna extension assembly 502 then operates to extend the petals of the antenna 302 when the antenna extension assembly 502 is moved toward the body 202 of the satellite 200. In one embodiment, a spring is coiled and exerting a spring force upon the antenna extension assembly 502, being held in place by a latch. Once the extruded state is reached, the latch is operated, thereby releasing the spring. The spring drives the antenna extension assembly 502 toward the body 202 of the satellite 200. The antenna extension assembly 502 then extends the petals of the antenna 302. The antenna extension assembly 502 may further come to rest against additional structure of the satellite 200.

In some embodiments, the satellite 200 does not reach a full extruded state prior to the transition to the extended state being initiated. The control system may at a point of partial extrusion, initiate the extension of the antenna 302 by the antenna extension assembly 502. In another embodiment, as the antenna 302 reaches a partially extruded state, a mechanical device, such as a latch may be operated by the movement of the antenna 302 in the extrusion transition to initiate the extension transition.

Referring to FIG. 7, an antenna 700 comprises a support structure 702, a petal 704-petal 718, S-band patches 720, a nadir panel 722, C-band patches 724, an extension plate 726, and an extension arm 728.

The support structure 702 is coupled to the body of the satellite. The support structure 702 may be attached with fasteners, welds, etc., may be press fit into the body of the satellite, or may have components that fit into grooves of the body of the satellite. The support structure 702 may be further coupled to other components of the antenna 700 to ensure the satellite remains integrated.

The petal 704-petal 718 are coupled to the nadir panel 722 and each to one extension arm 728. The petal 704-petal 718 may be rotatably coupled to each of the nadir panel 722 and the extension arm 728. As the extension plate 726 is moved toward the support structure 702 during the extension phase, force is exerted on each petal via the extension arm 728. The force is then applied to each rotatable joint causing each of the petals to align at a specified angle to the nadir panel 722 once the extension plate 726 has come to rest, e.g. between 5 and 45 degrees. The angle at which each of the petals is maintained contributes to the communication profile of the beams sent and received. As depicted in FIG. 7, the antenna 700 comprises eight (8) petals. In other embodiments, a different number of petals is utilized, such as six (6) petals. In many embodiments, the number of petals is an even number as the petals are grouped as send and receive pairs of petals. A send petal may have neighboring petals that are receive petals (and vice versa) to help ensure that a send petal and a receive petal is oriented toward a ground station. As the number of petals decreases, the antenna 700 is less likely to have an orientation to communicate with a ground station. Increasing the number of petals may increase the size of the antenna 700 (to maintain the same petal size) or decrease the size of each petal. Decreasing the size of each petal may then result in a reduced size of the S-band patches 720, which alters the beam profile for communication. The petal 704-petal 718 each are coupled to one or more S-band patches 720. The S-band patches 720 may be integrated into each of the petals, may be fastened to the petals, etc. As depicted, the antenna 700 has three (3) S-band patches 720 per petal. In some embodiments, each petal may have as few as one (1) patch or as many as eight (8) or more patches. Each of the petals (and patches) may be electrically coupled to a control system of the satellite. When a communication signal is received by a receive petal, the communication signal may be sent to the control system. The control system may then determine the destination of a send signal, select one of the send petals to send the signal, and send the communication signal to the selected send petal to emit the communication signal as part of its beam.

The nadir panel 722 is coupled to the support structure 702 and rotatably coupled to each of the petals. The nadir panel 722 may be coupled to the support structure 702 via intermediary structures (see FIG. 8). The nadir panel 722 may also be coupled to, or contact, an extension mechanism. Such an extension mechanism may also contact or be coupled to the extension plate 726. During the extension phase, the extension mechanism may operate to apply a force to each of the nadir panel 722 and the extension plate 726, causing those components to move away from each other. The nadir panel 722 is coupled to C-band patches 724. The C-band patches 724 may be integrated or fastened to the nadir panel 722. In some embodiments, two (2) C-band patches 724 are coupled to the nadir panel 722. One of the C-band patches 724 may be a send patch, while the other may be a receive patch. Each patch may be wired to a control system of the satellite. During operation, the nadir panel 722 is generally oriented toward the nadir.

The extension plate 726 is rotatably coupled to each extension arm 728. The extension plate 726 may also be slidably coupled to the support structure 702 or intermediary components (see FIG. 8). The extension plate 726 is further coupled or contacted to an extension mechanism. The extension mechanism, when activated, moves the extension plate 726 toward the support structure 702 and away from the nadir panel 722. The extension plate 726 exerts a force on each extension arm 728 via the rotatable coupling.

The extension arm 728 may be rotatably coupled to the extension plate 726 and one of the petals (as depicted, petal 704). In some embodiments, one extension arm 728 is coupled to one petal. In other embodiments, each extension arm 728 may be rotatably coupled to multiple petals. For example the extension arm 728 may be coupled to both petal 704 and the petal 706, resulting in an extension arm 728 with a Y-shape.

Referring to FIG. 8, an antenna 800 comprises a body support structure 802, an extruding mechanism 804, an extrusion plate 806, an extension plate 808, extension arms 810, an extension mechanism 812, an extrusion support structure 814, petals 816, and a nadir panel 818.

The body support structure 802 is coupled to the body of the satellite. The body support structure 802 is also coupled to the extruding mechanism 804. The body support structure 802 may be a hollow housing, such that, when the antenna 800 is in the stowed state, the other components are contained within the body support structure 802. The body support structure 802 may further be slidably coupled to the extrusion plate 806. The body support structure 802 may have a hollow cylindrical portion with an inner diameter that is similar to the

"Ground terminal" herein refers to a terrestrial radio station designed for extraplanetary telecommunication with the antenna.

"Horizon angle" herein refers to an angle measure from the horizon of the surface of an object from which the ground terminal transmits.

"Impedance transformer" herein refers to an electrical device that transfers electrical energy between two or more circuits through electromagnetic induction to match the impedances of the circuits.

"Motor" herein refers to a machine, which may be powered by electricity or internal combustion, that supplies motive power for a vehicle or for some other device with moving parts.

"One or more communication patches" herein refers to material utilized to send and receive signals utilizing the electromagnetic frequency spectrum.

"Receive petal" herein refers to a petal of a communication array with communication patches configured to receive communication signals.

"S-band patches" herein refers to material to communicate in the S-Band, utilizing the microwave band of the electromagnetic spectrum covering frequencies from 2 to 4 gigahertz (GHz).

"Send petal" herein refers to a petal of a communication array with communication patches configured to send communication signals.

"Transmission lines" herein refers to electrically conductive material utilize to electrically couple communication patches.

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